

DEPARTMENT OF TRANSPORTATION  
FEDERAL AVIATION ADMINISTRATION

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|---|
| E10EA-7<br>AVCO Lycoming<br>TIO-541-A1A<br>TIO-541-E1A4, -E1B4<br>-E1C4, -E1D4<br><br>April 4, 1986 |
|---|

TYPE CERTIFICATE DATA SHEET NO. E10EA

Engines of models described herein conforming with this data sheet (which is a part of Type Certificate No. E10EA) and other approved data on file with the Federal Aviation Administration, meet the minimum standards for use in certificated aircraft in accordance with pertinent aircraft data sheets and applicable portions of the Civil Air Regulations/Federal Aviation Regulations provided they are installed, operated and maintained as prescribed by the approved manufacturer's manuals and other approved instructions.

Type Certificate Holder                      AVCO Lycoming Williamsport Division  
AVCO Corporation  
Williamsport, Pennsylvania 17701

|   |                          |                            |
|---|--------------------------|----------------------------|
| Model    Lycoming TIO-541   | -A1A                     | -E1A4, -E1B4, -E1C4, -E1D4 |
| Type    6H0A Direct Drive Turbosupercharged   |                          |                            |
| Rating (See NOTE 4)   |                          |                            |
| Maximum continuous, hp., r.p.m. in. Hg., at:  | 310-2575-37.0-15,000     | 380-2900-41.9-15,000       |
| Standard density, critical alt. ft.   | 310-2575-37.0-S.L.       | 380-2900-41.0-S.L.         |
| Standard density, sea level alt. ft.  |                          |                            |
| Takeoff (5 min.), hp., r.p.m., in. Hg., at:   |                          |                            |
| Standard density, critical alt. ft.   | 310-2575-37.0-15,000     | 380-2900-41.9-15000        |
| Standard density sea level alt. ft.   | 310-2575-37.0-S.L.       | 380-2900-41.0-S.L.         |
| Fuel (minimum grade aviation gasoline)  | 100/130                  | --                         |
| Lubricating oil (lubricants should conform to the specification as listed or to subsequent revisions thereto) | Lycoming Spec. No. 301-E | --                         |
| Bore and stroke, in.  | 5.125 x 4.375            | --                         |
| Displacement, cu. in.   | 541.5                    | --                         |
| Compression Ratio   | 7.3:1                    | --                         |
| Weight (dry), lb.   | See NOTE 7               | --                         |
| C.G. location (with starter & generator installed)  | See NOTE 7               | --                         |
| Propeller shaft flange, SAE No. AS127   | Type 2 modified          | --                         |
| Crankshaft dampers (torsional)  | See NOTE 6               | --                         |
| Fuel injection  | See NOTE 7               | --                         |
| Turbosupercharger   | AiResearch (See NOTE 4)  | --                         |
| Ignition, dual  | Scintilla (See NOTE 7)   |                            |
| Ignition timing BTC   | 20                       | --                         |
| Spark plugs   | See NOTE 5               | --                         |
| Oil, sump capacity, qts. - total and usable   | See NOTE 7               | --                         |
| NOTES   | 1 thru 9                 | --                         |

"- -" indicates "same as preceding model"; "—" indicates "does not apply"

|          |   |   |   |
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*Reformatted 9/94.*

Certification basis:

| <u>Regulations and Amendments</u>       | <u>Model</u> | <u>Date of Application</u> | <u>Date Type Certificate<br/>No. E10EA Issued/Revised</u> |
|---|--------------|----------------------------|---|
| CAR 13 Effective June 15, 1956          | TIO-541-A1A  | December 16, 1963          | February 23, 1965   |
| As Amended By 13-1, 13-2, & 13-3 & 13-4 | TIO-541-E1A4 | December 14, 1965          | December 16, 1966   |
|   | TIO-541-E1B4 | June 1, 1966               | December 16, 1966   |
|   | TIO-541-E1C4 | October 13, 1969           | October 21, 1969  |
|   | TIO-541-E1D4 | October 13, 1969           | October 21, 1969  |

Production basis: Production Certificate No. 3

NOTE 1. Maximum permissible temperatures:

|  |   |
|--|---|
| Cylinder head (well type thermocouple)   | 475°F   |
| Cylinder base - This parameter dispensed with four engines equipped with internal piston cooling oil jets. See NOTE 8. |   |
| Oil inlet  | 245°F   |
| Fuel injector air inlet  | 300°F   |
| Exhaust gas (Turbo inlet at location shown on Lycoming Dwg. Nos. 63204   | 1650°F  |
| Compressor temperature rise  | (-A1A) -63250 (-E1A4, -E1B4) -63304 (-E1C4 & -E1D4))<br>(-A1A) 280°F (-E1A4, -E1B4, -E1C4, -E1D4) 320°F |

NOTE 2. Pressure limits.

|   | <u>Min.</u>  | <u>Max.</u>  | <u>Idle (Min.)</u> |
|---|--|--------------|--------------------|
| Fuel pressure limits<br>(above fuel injector inlet air press.)<br>(at inlet to fuel injector) | -A1A 29 p.s.i.<br>-E1A4, -E1B4<br>-E1C4, -E1D4 20 p.s.i. | 60 p.s.i.    | 12 p.s.i.          |
| Oil pressure limits:  | 60 p.s.i.  | 90 p.s.i.    | 25 p.s.i.          |
| Start and warm-up   | —  | 100 p.s.i.   | —                  |
| Fuel injector inlet pressure  | —  | 40.5 in. Hg. | —                  |
| Manifold pressure   | —  | 40 in. Hg.   | —                  |
| Turbo supercharger exhaust back pressure  | —  | 0.5 in. Hg.  | —                  |

NOTE 3. The following accessory provisions are available:

| Accessory                   | -A1A | -E1A4<br>-E1B4<br>-E1C4<br>-<br>E1D4 | Rotation Facing<br>Drive Pad | Speed Ratio<br>to Crankshaft | Maximum Torque<br>in. -lb.<br>Cont Static |       | Maximum Overhang<br>Moment<br>in. - lb. |
|-----------------------------|------|--------------------------------------|------------------------------|------------------------------|---|-------|---|
| Starter                     | *    | *                                    | CC                           | 16.566:1                     | —   | 450   | 150                                     |
| Alternator                  | *    | *                                    | C                            | 3.250:1                      | 60  | 120   | 175                                     |
| Vacuum or<br>Hydraulic Pump | *    | —                                    | C                            | 1.500:1                      | 200                                       | 1600  | 50                                      |
| Vacuum or<br>Hydraulic Pump | *    | —                                    | CC                           | 1.500:1                      | Total                                     | Total | 50                                      |
| Vacuum or<br>Hydraulic Pump | —    | *                                    | C                            | 1.000:1                      | 200                                       | 1600  | 50                                      |
| Vacuum or<br>Hydraulic Pump | —    | *                                    | CC                           | 1.000:1                      | Total                                     | Total | 50                                      |
| Tachometer                  | *    | *                                    | CC                           | .500:1                       | 7   | 50    | 5                                       |
| Propeller<br>Governor       | *    | *                                    | CC                           | .895:1                       | 125                                       | 825   | 25                                      |
| Fuel Pump                   | *    | *                                    | CC                           | 1.000:1                      | 25  | 450   | 25                                      |
| Air Compressor              | —    | *                                    | CC                           | 1.000:1                      | Belt Limited                              |       | 100                                     |

"C" - Clockwise, "CC" - Counter clockwise

\* Standard \*\* Optional

NOTE 4. These engines are equipped with an AiResearch turbosupercharger models mounted as an integral part of the engines as shown on Lycoming Drawings:

|              | Turbosupercharger<br>Model | Drawing No. |
|--------------|----------------------------|-------------|
| -A1A         | T-1823                     | 63204       |
| -E1A4, -E1B4 | T-1823                     | 63250       |
| -E1C4, -E1D4 | T-1879                     | 63304       |

Performance data for these engines are presented on Lycoming Curve Nos. 12987B(A1A), 13055A (E1A4, -E1B4, -E1C4, -E1D4).

This turbocharger meets the containment requirements of CAR 13.166 and does not require external protection. Air from the compressor of this turbocharger may be used for cabin pressurization.

NOTE 5. Spark plugs approved for use on these engines are listed in the latest revision of AVCO Lycoming Service Instruction No. 1042.

NOTE 6. These engines incorporate crankshafts with one fifth order and one sixth order dampers unless the digit "4" follows the model designation, i.e., -E1A4. Engines so designated, (---4), have one 3.5 order, one fourth order, one fifth order and one sixth order pendulum type counterweights.

NOTE 7. The following tabulation shows weight, C.G., fuel injectors, oil sump capacities and ignition:

| <u>Model</u> | <u>Weight</u> | <u>Center of Gravity, in.</u>                   |   | <u>Fuel<br/>Injector<br/>Bendix</u> | <u>Oil Sump Capacities (Qts)</u>            |      |                                 |
|--------------|---------------|---|---|-------------------------------------|---|------|---------------------------------|
|              |               | <u>From Front Face of<br/>Prop Shaft Flange</u> | <u>Off Centerline<br/>of Crankshaft</u> |                                     | <u>Usable 20°<br/>Total Nose up or down</u> |      | <u>Ignition Dual<br/>Bendix</u> |
| -A1A         | 549           | 21.00   | .06 Below<br>& .13 Right                | RSA-<br>10AD1                       | 14  | 10.0 | S6LN-1208;<br>S6RN-1209         |
| -E1A4        | 595           | 23.35   | .05 Below<br>& .05 Left                 | RSA-10DB1                           | 13  | 10.5 | S6LN-1208;<br>S6RN-1209         |
| -E1B4        | 595           | 23.35   | .05 Below<br>& .05 Left                 | RSA-10DB1                           | 13  | 10.5 | S6LN-1208;<br>S6RN-1209         |
| -E1C4        | 586           | 22.86   | .06 Below<br>& .03 Right                | RSA-10DB1                           | 13  | 10.5 | S6LN-1208;<br>S6RN-1209         |
| -E1D4        | 584           | 22.86   | .06 Below<br>& .03 Right                | RSA-10DB1                           | 13  | 10.5 | S6LN-1208;<br>S6RN-1209         |

NOTE 8. Model similarities and differences.

|             |  |
|-------------|--|
| TI0-541-A1A | Basic model. Six cylinder, air cooled, horizontally opposed, direct drive, fuel injection, turbosupercharged engine with topside induction, down exhaust and side mounted accessory drives. Provides for single acting, controllable pitch propeller and has internal piston cooling oil jets. |
| -E1A4       | Similar to -A1A but has higher rating, uses different cylinder heads, cam shaft, crankshaft and additional counterweights. Has cabin pressurizing venturi.   |
| -E1B4       | Similar to -E1A4 but does not incorporate a cabin pressurization venturi.  |
| -E1C4       | Similar to -E1A4 except for different turbosupercharger with cast bracket, cast transition and separate wastegate.   |
| -E1D4       | Similar to -E1B4 except for different turbosupercharger with cast bracket, cast transition and separate wastegate.   |

NOTE 9. Starters, generators and alternators approved for use on these engines are listed in the latest revision of AVCO Lycoming Service Instruction No. 1154.

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